

# 1           **Model Updating with Experimental Frequency Response**

## 2                           **Function Considering General Damping**

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7   **Abstract:** In order to obtain a more accurate finite element (FE) model of a constructed structure, a new frequency  
8 response function (FRF)-based model updating approach is proposed. A general viscous damping model is assumed in  
9 this approach for better simulating the actual structure. The approach is formulated as an optimization problem which  
10 intends to minimize the difference between analytical and experimental FRFs. Neither dynamic expansion nor model  
11 reduction is needed when not all degrees of freedom are measured. State-of-the-art optimization algorithms are utilized  
12 for solving the non-convex optimization problem. The effectiveness of the presented FRF model updating approach is  
13 validated through a laboratory experiment on a four-story shear-frame structure. To obtain the experimental FRFs, a  
14 shake table test was conducted. The proposed FRF model updating approach is shown to successfully update the stiffness,  
15 mass and damping parameters in matching the analytical FRFs with the experimental FRFs. In addition, the updating  
16 results are also verified by comparing time-domain experimental responses with the simulated responses from the updated  
17 model.

18

19   **Keywords:** frequency response function; finite element model updating; general viscous damping; non-convex  
20 optimization; shake table test

## 21

## 22

## 23   **1 Introduction**

24           With rapid development in numerical simulations, FE analysis has become a more and more  
25 powerful tool in structural engineering. Although significant improvements have been made towards  
26 accurate FE modeling, in general, there are still distinct differences between behaviors of a constructed  
27 structure and these of the FE model built according to the same design drawings. It is well known that

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28 analytical results from FE model often differ from performance of an actual structure in the field. The  
29 mismatch is mainly caused by nominal material property values, idealized boundary conditions,  
30 difficulties in modeling of damping, etc. To achieve an FE model that more accurately represents the  
31 actual structure, FE model updating can be performed through calibration with high-fidelity  
32 experimental test data. An accurate FE model can also be used later for structural safety monitoring and  
33 damage detection.

34 A number of FE model updating approaches have been proposed and practically applied during the  
35 past few decades, as reviewed by Imregun and Visser (1991). Friswell and Mottershead (1995)  
36 discussed detailed model updating techniques in their book. Most model updating approaches can be  
37 broadly categorized into time domain approaches and frequency domain approaches. Time domain  
38 approaches usually use vibration data to directly update the FE model (Yang *et al.* 2009; Hernandez and  
39 Bernal 2013). Although they have merits, the computational efforts are usually a concern. On the other  
40 hand, in frequency domain, most approaches need to use the experimental modal properties of a  
41 structure to construct an optimization problem for model updating (Zhu *et al.* 2016; Brito *et al.* 2014;  
42 Jaishi and Ren 2005; Tshilidzi and Sibusiso 2005). The optimization problem generally attempts to  
43 minimize an objective function that contains the difference between experimental and simulated natural  
44 frequencies, mode shapes and modal flexibilities, etc. In these model updating approaches, extraction  
45 of modal properties from experimental data is first required, which can add uncertainties and  
46 inaccuracies to the updating. In addition, in most cases, only limited amount of modal information can  
47 be obtained from modal analysis. As summarized by Jaishi and Ren (2005), an accurate model can be  
48 achieved only when the number of extracted experimental modal properties is greater than or equal to  
49 the number of interested updating variables.

50 This research focuses on another category in frequency domain model updating approaches, which  
51 is based on frequency response functions (FRF). From experimental data, FRFs can be easily calculated  
52 using excitation record and corresponding structural responses. This avoids the need for extracting  
53 modal properties and the associated extraction errors. Furthermore, high quality FRFs can be obtained  
54 by using FRF estimators to minimize the influence of noise in the calculation (Schoukens and Pintelon

55 1990; Antoni *et al.* 2004). Another advantage of FRF-based approaches is that an experimental test can  
56 provide abundant FRF data in a large frequency range. Owing to these advantages, FRF-based  
57 approaches constitute a highly valuable category in FE model updating.

58 Among the most widely known FRF-based model updating approaches was proposed by Lin and  
59 Ewins (1994), which avoids the inverse of the system dynamic stiffness matrix by using the analytical  
60 FRF sensitivity matrix. This approach usually can perform accurately and efficiently on numerical  
61 simulation cases, because of the assumptions of noise-free and complete measurements on all degrees  
62 of freedoms (DOFs). However, such assumptions, particularly the sensor instrumentation on all DOFs,  
63 are usually unrealistic in practice. Through model reduction technique, Asma and Bouazzouni (2005)  
64 later extended Lin and Ewins' work to update a truss structure with incomplete measurement.  
65 Alternatively, Avitabile and O'callahan (2001) presented the dynamic expansion approach to get a full  
66 column or row of an FRF matrix. Nevertheless, it is well known that neither reduction nor expansion  
67 can fully describe the actual dynamic behavior of a structure. To overcome this limitation, Sipple and  
68 Sanayei (2014) proposed a numerically evaluated FRF sensitivity-based model updating approach.  
69 Optimization techniques are utilized to iteratively change the analytical FRFs to match the experimental  
70 counterparts. The modal-decomposed analytical FRF is in scalar form which can be directly used in  
71 updating process. The model reduction or dynamic expansion is not necessary in this case.

72 Due to the existence of damping, the experimental FRFs are usually complex valued functions.  
73 Despite the large amount of literature on damping modeling, damping still remains the least known  
74 aspect compared with stiffness and mass. In order to avoid the difficulties in damping updating, Pradhan  
75 and Modak (2012) proposed to use the real-valued normal FRF matrix  $(-\omega^2\mathbf{M} + \mathbf{K})$  in model updating.  
76 However, when formulating the estimation of the normal FRFs, the method requires the full complex-  
77 valued FRF matrix which has to be estimated through the identified modal properties. The estimation  
78 may still require modal identification and add inaccuracies. Since damping cannot be ignored in  
79 practical modeling, especially with complex FRFs, a proper selection of damping model may improve  
80 the model updating accuracy. Among all damping models, viscous damping is the most commonly used  
81 due to its convenience in structural design. Another model, hysteretic damping, can more accurately

82 describe the energy dissipation in structure vibration, the difficulty of translating this damping  
83 mechanism into time domain prevents an easy adoption. In addition, Lim and Zhu (2009) demonstrated  
84 that the difference caused by arbitrarily choosing hysteretic damping and viscous damping in system  
85 identification is small. Therefore, most researchers prefer to assume proportional viscous damping (i.e.  
86 Rayleigh damping or Caughey damping) for FRF-based model updating. For example, Imregun *et al.*  
87 (1995) and Hong *et al.* (2016) updated the Rayleigh damping coefficients through an extended FRF-  
88 based model updating approach. Lu and Tu (2004), Sipple and Sanayei (2014) updated the damping  
89 ratios (corresponding to Caughey damping) in their FRF-based model updating. Nevertheless,  
90 proportional damping may rarely exist in reality, and most structures possess non-proportional damping.  
91 From this point of view, the use of proportional damping will more or less affect the updating accuracy.  
92 A general viscous damping model (which includes both proportional damping and non-proportional  
93 damping) can render more accurate model updating.

94 This research departs from the authors' preliminary study (Hong *et al.* 2016). We focus on a model  
95 updating approach that can minimize the difference between analytical and experimental FRFs directly  
96 at measured DOFs. This differs from most FRF-based model updating approaches in literature that need  
97 reduction or expansion techniques. In comparison with Sipple and Sanayei (2014) and Hong *et al.*  
98 (2016), a general viscous damping assumption is provided for better simulating actual structures in  
99 reality. The FRF formulation is derived for a base excitation setup when ground vibration occurs to a  
100 shear-frame building structure (which effectively applies excitation simultaneously at all DOFs). To  
101 validate the proposed FRF-based model updating, shake table tests are performed on a four-story  
102 laboratory structure in this study, although the authors are currently extending the formulation for future  
103 application to a space frame bridge. The rest of the paper is organized as follows. First, Section 2  
104 presents the analytical FRF formulation for a structure undergoing ground excitation and the  
105 experimental FRF calculation. In section 3, the vector form of the analytical FRF to be used in model  
106 updating is introduced; then the optimization procedure is discussed. Section 4 describes the shake table  
107 test on a four-story aluminum structure for validating the performance of the proposed formulations for  
108 FRF-based model updating. We compare the experimentally measured frequency domain FRFs and

109 time domain response histories with their counterparts simulated using the updated model. Finally,  
 110 conclusions and future work are provided in Section 5.

111

## 112 **2 Formulations of the frequency response functions**

### 113 *2.1 Analytical formulation of the frequency response functions considering general viscous damping*

114 Consider the dynamic equation of motion of an  $n$ -DOF structure with viscous damping at time  $t$ :

$$115 \quad \mathbf{M}\ddot{\mathbf{q}}(t) + \mathbf{C}\dot{\mathbf{q}}(t) + \mathbf{K}\mathbf{q}(t) = \mathbf{F}(t) \quad (1)$$

115 where  $\mathbf{M}, \mathbf{K}, \mathbf{C} \in \mathbb{R}^{n \times n}$  are mass, stiffness and damping matrices, respectively;  $\mathbf{q} \in \mathbb{R}^n$  is the  
 116 displacement vector;  $\mathbf{F} \in \mathbb{R}^n$  is the force vector.

117 To decompose structural response with non-proportional damping, a strategy is to rewrite Eqn 1 in  
 118 state space, so that the  $n$  number of second-order differential equations can be converted to  $2n$  number  
 119 of first-order differential equations.

$$120 \quad \mathbf{A}\dot{\mathbf{x}}(t) + \mathbf{B}\mathbf{x}(t) = \mathbf{P}(t) \quad (2)$$

120 where  $\mathbf{x} \in \mathbb{R}^{2n}$  is the state vector. In order to make Eqns 1 and 2 equivalent,  $\mathbf{x}, \mathbf{A}, \mathbf{B}, \mathbf{P}$  are defined as  
 121 follows,

$$122 \quad \mathbf{x}(t) = \begin{Bmatrix} \mathbf{q}(t) \\ \dot{\mathbf{q}}(t) \end{Bmatrix}_{2n \times 1} \quad (3)$$

$$123 \quad \mathbf{A} = \begin{bmatrix} \mathbf{C} & \mathbf{M} \\ \mathbf{M} & \mathbf{0} \end{bmatrix}_{2n \times 2n}, \mathbf{B} = \begin{bmatrix} \mathbf{K} & \mathbf{0} \\ \mathbf{0} & -\mathbf{M} \end{bmatrix}_{2n \times 2n}, \mathbf{P}(t) = \begin{bmatrix} \mathbf{F}(t) \\ \mathbf{0} \end{bmatrix}_{2n \times 1} \quad (4)$$

122 Complex eigenvalues  $s_i \in \mathbb{C}$  and eigenvectors  $\boldsymbol{\psi}_i \in \mathbb{C}^{2n}$  ( $i = 1, 2, \dots, 2n$ ) can be obtained by  
 123 solving the generalized eigenvalue problem of the state-space system,

$$124 \quad (s_i \mathbf{A} + \mathbf{B})\boldsymbol{\psi}_i = \mathbf{0}, \quad i = 1, 2, \dots, 2n \quad (5)$$

124 where  $\boldsymbol{\psi}_i$  ( $i = 1, 2, \dots, 2n$ ) are the eigenvectors normalized with respect to  $\mathbf{A}$  matrix, i.e.,  $\boldsymbol{\Psi}^T \mathbf{A} \boldsymbol{\Psi} =$   
 125  $\mathbf{I}_{2n \times 2n}$  with the eigenvector matrix defined as  $\boldsymbol{\Psi} = [\boldsymbol{\psi}_1 \quad \boldsymbol{\psi}_2 \quad \dots \quad \boldsymbol{\psi}_{2n}] \in \mathbb{C}^{2n \times 2n}$ . The superscript ‘T’  
 126 represents matrix transpose. As a result, denoting the diagonal eigenvalue matrix  $\mathbf{S} =$

127  $\text{diag}(s_1, s_2, \dots, s_{2n}) \in \mathbb{C}^{2n \times 2n}$ , we have  $\Psi^T \mathbf{B} \Psi = -\mathbf{S}$ . It is also well known that the complex-valued  
 128 eigenvector  $\boldsymbol{\psi}_i$  can be expressed as

$$\boldsymbol{\psi}_i = \begin{Bmatrix} \boldsymbol{\phi}_i \\ s_i \boldsymbol{\phi}_i \end{Bmatrix} \quad (6)$$

129 where  $\boldsymbol{\phi}_i$  is an  $n \times 1$  complex vector, which represents the modal displacements. Defining  $\mathbf{z}(t) \in \mathbb{C}^n$   
 130 as the modal coordinate vector, the relationship between the state vector and modal coordinate vector is  
 131 shown below,

$$\mathbf{x}(t) = \Psi \mathbf{z}(t) \quad (7)$$

132 Substituting Eqn 7 into Eqn 2, we get

$$\mathbf{A} \Psi \dot{\mathbf{z}}(t) + \mathbf{B} \Psi \mathbf{z}(t) = \mathbf{P}(t) \quad (8)$$

133 Pre-multiplying Eqn 8 by  $\Psi^T$  results in

$$\dot{\mathbf{z}}(t) - \mathbf{S} \mathbf{z}(t) = \Psi^T \mathbf{P}(t) \quad (9)$$

134 Because  $\mathbf{S}$  is a diagonal matrix, Eqn 9 in vector form can be easily decoupled into  $2n$  number of  
 135 scalar differential equations. Recalling  $\boldsymbol{\psi}_i = \begin{Bmatrix} \boldsymbol{\phi}_i \\ s_i \boldsymbol{\phi}_i \end{Bmatrix}$  and  $\mathbf{P}(t) = \begin{bmatrix} \mathbf{F}(t) \\ \mathbf{0} \end{bmatrix}$ , we get

$$\dot{z}_i(t) - s_i z_i(t) = \boldsymbol{\phi}_i^T \mathbf{F}(t), \quad i = 1, 2, \dots, 2n \quad (10)$$

136 Furthermore, through Fourier transform, Eqn 10 can be expressed in frequency domain as,

$$j\omega \hat{z}_i(\omega) - s_i \hat{z}_i(\omega) = \boldsymbol{\phi}_i^T \hat{\mathbf{F}}(\omega) \quad (11)$$

137 where  $j = \sqrt{-1}$  is the imaginary unit;  $\omega$  represents frequency. Using  $\mathcal{F}\{\cdot\}$  to represent Fourier  
 138 transform,  $\hat{z}_i = \mathcal{F}\{z_i\}$  is the  $i$ -th modal coordinate and  $\hat{\mathbf{F}} = \mathcal{F}\{\mathbf{F}\}$  is the force vector in frequency  
 139 domain.

140 Then, collect the terms in Eqn 11 and express the modal coordinate as

$$\hat{z}_i(\omega) = \frac{\boldsymbol{\phi}_i^T \hat{\mathbf{F}}(\omega)}{j\omega - s_i} \quad (12)$$

141 In order to find the relationship between the input force and output displacement, we transform  
 142 Eqn 7 into frequency domain and then substitute Eqn 12 into it,

$$\hat{\mathbf{x}}(\omega) = \boldsymbol{\Psi} \hat{\mathbf{z}}(\omega) = \sum_{i=1}^{2n} \boldsymbol{\psi}_i \hat{z}_i(\omega) = \sum_{i=1}^{2n} \frac{\boldsymbol{\psi}_i \boldsymbol{\phi}_i^T \hat{\mathbf{F}}(\omega)}{j\omega - s_i} \quad (13)$$

143 As shown in Eqn 3, the upper half of the state vector corresponds to displacement. So the frequency  
 144 domain displacement  $\hat{\mathbf{q}}$  can be expressed as below,

$$\hat{\mathbf{q}}(\omega) = \sum_{i=1}^{2n} \frac{\boldsymbol{\phi}_i \boldsymbol{\phi}_i^T \hat{\mathbf{F}}(\omega)}{j\omega - s_i} = \mathbf{H}(\omega) \hat{\mathbf{F}}(\omega) \quad (14)$$

145 Base on the derivation,  $\mathbf{H} = \sum_{i=1}^{2n} \frac{\boldsymbol{\phi}_i \boldsymbol{\phi}_i^T}{j\omega - s_i} \in \mathbb{C}^{n \times n}$  is the receptance (displacement) FRF matrix,  
 146 which represents the mapping from force input to displacement output.

147 Eqn 15 shows the  $(r, e)$  entry in the receptance matrix, which represents the input-output  
 148 relationship from excitation at the  $e$ -th DOF to the response at the  $r$ -th DOF.

$$H_{r,e}(\omega) = \sum_{i=1}^{2n} \frac{\phi_{r,i} \phi_{e,i}}{j\omega - s_i} \quad (15)$$

149 where  $\phi_{r,i}$  and  $\phi_{e,i}$  are the  $r$ -th and  $e$ -th entry of the  $i$ -th complex modal displacement vector  $\boldsymbol{\phi}_i$ ,  
 150 respectively.

151 To derive the FRFs from ground excitation to structural response, a similar approach is adopted as  
 152 used by Hong *et al.* (2016). For an  $n$ -DOF shear-frame structure, the displacement at DOF- $r$  caused by  
 153 ground acceleration ( $A_g$ ) is calculated as the summation of all contributions to displacement at DOF- $r$   
 154 caused by the equivalent earthquake forces at all DOFs. Therefore, the analytical form of the FRFs with  
 155 ground excitation can be extended from Eqn 15. Let  $X_{r,e}(\omega)$  represent displacements at DOF- $r$  due to  
 156  $F_e(\omega)$ , the excitation at DOF- $e$  in frequency domain; and  $m_e$  be the lumped mass at DOF- $e$ .

$$X_{r,g}(\omega) = \sum_{e=1}^n X_{r,e}(\omega) = \sum_{e=1}^n H_{r,e}(\omega) F_e(\omega) = \sum_{e=1}^n -H_{r,e}(\omega) A_g(\omega) m_e \quad (16)$$

157 The receptance for response at location  $r$  due to ground excitation can be derived from Eqn 16 :

$$H_{r,g}(\omega) = \frac{X_{r,g}(\omega)}{A_g(\omega)} = \sum_{e=1}^n -H_{r,e}(\omega) m_e = \sum_{i=1}^{2n} \frac{-\phi_{r,i} \sum_{e=1}^n m_e \phi_{e,i}}{j\omega - s_i} \quad (17)$$

158 For other types of measurement data besides displacement, FRF formulation for ground excitation  
159 can be easily changed to other forms. These include the mobility,  $Y_{r,g}(\omega)$ , which represents the velocity  
160 response, and acceleration,  $A_{r,g}(\omega)$ , which represents the acceleration response.

$$Y_{r,g}(\omega) = j\omega H_{r,g}(\omega) \quad (18)$$

$$A_{r,g}(\omega) = -\omega^2 H_{r,g}(\omega) \quad (19)$$

161

## 162 2.2 Calculation of frequency response function from experimental data

163 In a large number of literatures related to FE model updating through FRFs, researchers only  
164 devoted their efforts in analytical FRF formulation rather than on how to calculate the experimental FRF  
165 through test data. It is well known that if the signal is polluted by noise, the model updating results are  
166 easily affected. However, measurement noise is impossible to avoid in experimental testing, which calls  
167 for the need of advanced FRF estimators to calculate the experimental FRFs. Researchers can choose  
168 different estimators to calculate the experimental FRF depending on their needs. For convenience,  $H_1$   
169 estimator (Schoukens and Pintelon 1990), as one of the most commonly used, is adopted here.

$$H_1(\omega) = \frac{S_{xy}(\omega)}{S_{xx}(\omega)} \quad (20)$$

170 where  $S_{xy}$  is the cross-spectral density between the excitation force and response signal;  $S_{xx}$  is the auto-  
171 spectral density of the response signal.

172

## 173 **3 Frequency response function-based model updating approach and optimization** 174 **procedures**

### 175 3.1 Analytical vector form of frequency response functions for model updating

176 Because of the damping effect, the FRF formulations in Eqns 17 to 19 are complex-valued.  
177 Experiences suggest that the use of FRF magnitude (i.e.  $\bar{H}_{r,g}(\omega)$ ,  $\bar{A}_{r,g}(\omega)$ ) provide better results than  
178 the use of either the real part or the imaginary part of FRF or their combinations in the model updating  
179 process.



$$\bar{H}_{r,g}(\omega) = |H_{r,g}(\omega)| \quad (21)$$

180 The expression in Eqn 21 is a scalar representing magnitude evaluated at frequency  $\omega$ . To include  
 181 FRF values at multiple frequency points of interest, as well as FRFs from multiple response DOFs, a  
 182 long vector is formulated as below,

$$\mathbf{H}^{\text{Ana}} = \left\{ [\bar{H}_{r_a,g}(\omega_1), \dots, \bar{H}_{r_a,g}(\omega_{n_\omega})], \dots, [\bar{H}_{r_p,g}(\omega_1), \dots, \bar{H}_{r_p,g}(\omega_{n_\omega})] \right\}^T \quad (22)$$

183 The subscripts  $r_a, \dots, r_p$  represent different response DOFs.  $\omega_i$  ( $i=1, \dots, n_\omega$ ) is an interested  
 184 frequency point, an  $n_\omega$  is the total number of interested frequency points.

185 The main objective of the model updating is to minimize the difference between analytical and  
 186 experimental FRFs.  $\mathbf{H}^{\text{Ana}}$  is the long analytical FRF vector, and  $\mathbf{H}^{\text{Exp}}$  is the experimental counterpart.  
 187 Both  $\mathbf{H}^{\text{Exp}}$  and  $\mathbf{H}^{\text{Ana}}$  should be written in the same sequence as show in Eqn 22.

188

### 189 3.2 Frequency response function-based model updating

190 For a linear structure, the stiffness and mass matrices can be expressed as matrix functions of the  
 191 updating variables  $\boldsymbol{\alpha} \in \mathbb{R}^{n_\alpha}$  and  $\boldsymbol{\beta} \in \mathbb{R}^{n_\beta}$ , respectively. Notation  $n_\alpha$  and  $n_\beta$  represent the total number of  
 192 updating variables associated with stiffness and mass, respectively; the  $r$ -th entry of  $\boldsymbol{\alpha}$  and  $\boldsymbol{\beta}$ ,  $\alpha_r$  and  
 193  $\beta_r$  are the relative change percentage associated with physical parameters to be updated, such as  
 194 Young's modulus, support spring stiffness and mass density, etc.

$$\mathbf{K} = \mathbf{K}_0 + \sum_{r=1}^{n_\alpha} \alpha_r \mathbf{K}_{0,r} \quad (23)$$

$$\mathbf{M} = \mathbf{M}_0 + \sum_{r=1}^{n_\beta} \beta_r \mathbf{M}_{0,r} \quad (24)$$

195 where  $\mathbf{K}_0$  and  $\mathbf{M}_0$  are the constant nominal stiffness and mass matrices, respectively, as the starting point  
 196 of the modeling;  $\mathbf{K}_{0,r}$  and  $\mathbf{M}_{0,r}$  are the constant influence matrices which corresponding to  $\alpha_r$  and  $\beta_r$ ,  
 197 respectively.

198 In this research, general viscous damping is assumed for simulating more practical structural  
 199 system. As mentioned by Chopra (2001), it is impractical to build the damping matrix in the form of

200 building stiffness matrix. Therefore, every entry in the damping matrix can be defined as an optimization  
 201 variable. In order to reduce the number of the damping updating variables, the damping matrix is  
 202 constrained to be symmetric positive definite (denoted as  $\mathbf{C} > 0$ ).

203 The complete optimization problem is provided as follows.

$$\text{minimize } \|\mathbf{H}^{\text{Ana}}(\boldsymbol{\alpha}, \boldsymbol{\beta}, \mathbf{C}) - \mathbf{H}^{\text{Exp}}\|^2 \quad (25a)$$

$$\text{subject to } (s_i \mathbf{A} + \mathbf{B})\boldsymbol{\psi}_i = \mathbf{0} \quad (25b)$$

$$\mathbf{H}^{\text{Ana}} = \left\{ [\bar{H}_{r,a,g}(\omega_1), \dots, \bar{H}_{r,a,g}(\omega_{n_\omega})], \dots, [\bar{H}_{r,p,g}(\omega_1), \dots, \bar{H}_{r,p,g}(\omega_{n_\omega})] \right\}^T \quad (25c)$$

$$H_{r,g}(\omega) = \sum_{i=1}^m \frac{-\phi_{r,i} \sum_{e=1}^n m_e \phi_{e,i}}{j\omega - s_i} \quad (25d)$$

$$\boldsymbol{\alpha}^l \leq \boldsymbol{\alpha} \leq \boldsymbol{\alpha}^u; \boldsymbol{\beta}^l \leq \boldsymbol{\beta} \leq \boldsymbol{\beta}^u \quad (25e)$$

$$\mathbf{C} > 0 \quad (25f)$$

204 where  $\|\cdot\|$  can be any norm function;  $\boldsymbol{\alpha}$ ,  $\boldsymbol{\beta}$ ,  $\mathbf{C}$  are the selected optimization variables. Lower bound  
 205 (superscript l) and upper bound (superscript u) are set for those updating variables corresponding to  
 206 physical parameters.  $m$  is the total number of analytical modes used in model updating.

207

### 208 3.3 Optimization procedures

209 The frequency response function-based model updating approach is formulated as a constrained  
 210 optimization problem in Eqn 25. There are several optimization algorithms that can be utilized for  
 211 finding the optimum value for the variables, such as nonlinear least-square, particle swarm and Newton  
 212 method, etc. In this research, a constrained nonlinear multivariable function solver ‘fmincon’ in  
 213 MATLAB optimization toolbox (Math Works Inc. 2015) is adopted for solving the problem. In general,  
 214 the optimization problem in Eqn 25 is non-convex and there is no optimization algorithm can guarantee  
 215 the global optimality of the solution. In order to increase the possibility of finding the global optimal  
 216 value for the problem, ‘Global Search’ in MATLAB is recommended to use together with ‘fmincon’.

217

#### 217 3.3.1 ‘fmincon’ in MATLAB

218           The solver ‘fmincon’ seeks a minimum of the objective function value to match the analytical FRFs  
219 with the experimental FRFs. One of the many advantages is that both equality and inequality constraints  
220 are allowed in this solver. In addition, the lower and upper bounds for optimization variables are  
221 allowed. Four algorithms are implemented in ‘fmincon’ optimization solver, including the trust region  
222 reflective algorithm, active set algorithm, sequential quadratic programming (SQP) algorithm and  
223 interior-point algorithm. Among them, the trust region reflective algorithm needs to provide gradient  
224 information of the objective function by the user. From this point of view, the algorithm is not suitable  
225 for those objective functions whose gradients are difficult to explicitly write in closed form. Other than  
226 this limitation, the trust region reflective algorithm allows user to set either bounds or linear equality  
227 constraints. Active set algorithm and SQP algorithm are not suitable for large scale problem. Since the  
228 interior-point algorithm does not have obvious drawbacks, it is adopted as the first trial. The interior-  
229 point algorithm is essentially a quasi-Newton method, which calculates the Hessian approximation by  
230 the BFGS algorithm. The interior-point algorithm will first attempt a direct-step to solve the KKT  
231 conditions; if unsuccessful, a conjugate-gradient search will be adopted instead. When numerically  
232 evaluating the gradient, based on the author’s experience, the minimum change of updating variables  
233 (‘DiffMinChange’ option) can be set comparatively larger for a highly nonlinear optimization problem.  
234 Allowing larger minimum change makes the gradient calculation more robust against inaccurate  
235 objective function evaluations due to numerical noises.

### 236           3.3.2 ‘Global Search’ in MATLAB

237           Because of the non-convexity of the objective function, the ‘fmincon’ solver may easily get trapped  
238 into a local minimum or even stop near the initial starting point. In order to increase the chance to find  
239 a more optimal solution for the objective function, the optimization procedure can be started from many  
240 initial points. ‘Global Search’ in MATLAB as a global optimization toolbox can help generate many  
241 initial points for local solvers using a scatter-search algorithm. It analyzes the initial points and only  
242 accept those points who can improve the optimization results. The drawback of ‘Global Search’ in  
243 MATLAB is that it can only run together with local solver ‘fmincon’. The number of starting points can  
244 be set by experience. The more starting points one uses, the higher the chance is in finding a better

245 solution with a smaller objective function value. On the other hand, more starting points usually  
246 consume more computing time.

247

## 248 **4 Experimental validation**

249 In this section, the performance of the FRF-based model updating approach is validated through a  
250 four-story shear-frame laboratory structure. How to select the frequency points for matching the FRFs  
251 is also discussed.

### 252 *4.1 Shake table (ground excitation) test*

253 The test structure shown in Figure 1 is mounted on a small shake table. All the column bars and  
254 floor plates are made of the same aluminum material. Every floor plate has the same mass 4.64kg. As  
255 initial starting point for mass variables, this number does not include the mass of sensor instrumentation  
256 on each floor; the model updating is expected to update the total mass so that equivalently the  
257 instrumentation mass is identified through updating. Every story has 8 thin column bars riveted to the  
258 plate. The rectangular section is  $0.0254\text{m} \times 0.00159\text{m}$ . The Young's modulus of the material is 63GPa.  
259 The total height of the structure is 1.182m ( $0.305\text{m} \times 3 + 0.267\text{m}$ ). Fixed connections are applied at the  
260 bottom of the every column. This structure can be idealized as a 4-DOF system since every floor can be  
261 taken as a rigid mass, and the lateral stiffness are mainly provided by bending of the columns. The  
262 model updating is expected to identify the inter-story shear stiffness provided by the columns. There are  
263 in total 5 accelerometers and 5 linear variable differential transformers (LVDT) installed on the structure  
264 for measuring the vibration. The accelerometer and LVDT on the same floor are interfaced with one  
265 wireless sensing system, *Martlet* (Kane *et al.* 2014). More detailed descriptions of the structure and  
266 sensors can be found in Hong *et al.* (2016).

267 During the shake table test, the ground earthquake is simulated by a chirp excitation which changes  
268 from 0Hz to 10Hz within 60s. The sampling frequency of the *Martlet* is set to be 200Hz. Figure 2 shows  
269 the measured ground excitation time history. To get enough FRF data for model updating, the  
270 experimental accelerances and receptances are calculated using the measured acceleration and

271 displacement response on every floor with the ground excitation, respectively. Figure 3 shows  
272 acceleration and displacement responses of the 4<sup>th</sup> floor.

#### 273 *4.2 Frequency points selection*

274 Figure 4 and figure 5 show the overlay plots of accelerances and receptances in frequency domain,  
275 respectively. In order to illustrate the resonance areas more clearly, all FRFs are plotted in dB form.  
276 There are 4 obvious peaks which correspond to 4 resonant frequencies. Although a large number of FRF  
277 points can be obtained from these curves, it is not recommended to use all frequency points for model  
278 updating. First, we notice the regions away from resonances are not as clean as the resonant areas,  
279 because the influence of sensor noise is more predominant at the regions with low energy near anti-  
280 resonances. FRF data in such regions with low signal-to-noise ratio (SNR) negatively affect the model  
281 updating accuracy, and thus, should not be used for matching with the analytical FRFs towards model  
282 updating. Since damping parameters in this structure are important optimization variables and damping  
283 effect mainly manifests around the resonances, the peak areas of each FRF curve are chosen as the target  
284 for matching the analytical FRFs. From our experience, half-power bandwidth method is recommended  
285 to identify the target frequency points around each resonance. Esfandiari *et al.* (2016) mentioned the  
286 importance of using FRF data in high frequency range for model updating, because high frequency  
287 corresponds to local structural vibration patterns. Therefore, it would be better to include the 4<sup>th</sup> peak  
288 (although they have relatively low amplitude) for model updating. In this study, the FRF calculated from  
289 responses on all floors will be used for a updating, although the updating can still be performed using  
290 data from only some floors.

#### 291 *4.3 Model updating result*

292 For this 4-story structure, the optimization variables include the inter-story shear stiffness of each  
293 floor, the mass of each floor, and each entry of the damping matrix. Mainly contributed by shear stiffness  
294 from the fixed-end columns, the initial story stiffness values are calculated based on the nominal  
295 Young's modulus of the material and the fix-end assumptions. The initial mass value is the 4.64 kg plate  
296 mass. The reason to choose the mass of each floor as updating variables is that the mass of sensor  
297 instrumentation cannot be neglected on this laboratory-scale structure. It is easy to find that mass and

298 stiffness information cannot be all updated through most modal property-based updating approaches,  
299 because of the scaling effect to stiffness and mass in the eigenvalue equation. Unlike these modal  
300 property-based updating approaches, the use of eigenvectors normalized with respect to  $\mathbf{A}$  matrix (Eqn  
301 5) prevents the scaling effect, allowing us to update all mass and stiffness values simultaneously. The  
302 lower bounds and upper bounds for mass and stiffness allow the variables to change in a reasonable  
303 range.

304 Table 1 summarizes the model updating results for the variables related to mass and stiffness.  
305 Analytical receptances and accelerances are updated through Eqn 25, respectively. In the last row of  
306 Table 1, the average updated values of mass and stiffness variables are calculated. Since damping  
307 updating is most difficult, we set the initial starting damping matrix as a Rayleigh damping matrix. The  
308 Rayleigh damping coefficients are chosen based on experience. In addition, during the updating process,  
309 the lower bounds and upper bounds for damping updating variables are set to be relatively large.

310 Figure 6 shows an example of the updating FRF plots using the proposed model updating approach.  
311 Figure 6(a) compares the initial, the experimental and updated accelerance  $\mathbf{A}_{3,g}$ . Figure 6(b) shows the  
312 comparison for receptance  $\mathbf{H}_{3,g}$ . The comparison plots demonstrate that the proposed approach is able  
313 to well match analytical FRFs with experimental ones. In particular, the peak areas for these updated  
314 FRF curves can match well with the peak areas of experimental FRF curves. Because damping controls  
315 the amplitude of the FRF at frequency points close to resonances, this result shows damping of the  
316 structure is updated with good accuracy. The frequency domain assurance criterion (FDAC) (Pascual *et al.*  
317 1997) value can be utilized to compare the similarity of the peak areas between the updated and  
318 experimental FRFs. A value 1 means perfect correlation, 0 means no correlation at all. The FDAC value  
319 in Figure 6(a) is 0.987 and the FDAC value in Figure 6(b) is 0.991.

320 In order to further verify the model updating results, a time domain comparison is also conducted.  
321 The average value of each optimization variable is used for building a new analytical model; the  
322 measured ground acceleration is fed into the model for simulating dynamic responses. Figure 7(a) shows  
323 an overall comparison between the simulated acceleration (from the new model) and the experimental  
324 acceleration on the 4<sup>th</sup> floor. Figure 7(b) is a close-up comparison for a three-second duration with the

325 highest amplitude, demonstrating a close match between simulated and experimental time histories. In  
326 addition, Figure 8(a) shows the overall comparison between the simulated displacement from the new  
327 model and the experimental displacement on the same floor. Figure 8(b) also gives a close-up  
328 comparison. All figures illustrate excellent agreement between the simulated results and the  
329 experimental data, which demonstrates the ability of the proposed FRF-based model updating approach  
330 in obtaining an accurate FE model to represent the structure.

#### 331 *4.4 Performance of the optimization toolbox*

332 As discussed in section 3.3, ‘fmincon’ and ‘Global Search’ in MATLAB toolbox are shown to be  
333 suitable for solving the optimization problem in this study. One of the biggest advantages is the  
334 simplicity for implementation. For this research, the convergence limits for objective function value and  
335 each optimization variable are set as  $10^{-6}$  and  $10^{-8}$ , respectively. In order to achieve more optimal  
336 updating results, a comparatively large number (10,000) of trial starting points are adopted for ‘Global  
337 Search’. Although more trial points mean higher time consumption, the inherent scatter search  
338 algorithm automatically eliminates the less promising starting points, effectively reducing the  
339 computation. The results shown in section 4.3 indicate good performance of the optimization toolbox  
340 for updating the 4-story structure. However, for more complex structures, the non-convexity of the  
341 objective function may be more significant, thus the optimization difficulty can increase accordingly.

342

### 343 **5 Summary and future work**

344 A summary of this work is first provided as follows:

345 1) The proposed FRF-based model updating approach has been investigated through a laboratory  
346 structure. In order to consider general viscous damping, the analytical formulation of FRF was derived  
347 in state space. Unlike other FRF-based model updating approaches, the proposed approach does not  
348 require the analytical FRF sensitivity matrix (which is impossible or difficult to get in most cases) for  
349 each updating variable. No model reduction or modal expansion is needed.

350 2) The proposed model updating approach can be easily implemented using state-of-the-art  
351 optimization toolboxes. MATLAB optimization solvers ‘fmincon’ and ‘Global Search’ have been

352 carefully discussed. Leveraging these optimization techniques, it is more likely to find an objective  
353 function value closer to the global minimal for the non-convex problem.

354 3) The proposed approach was successfully applied on the model updating of a 4-story structure.  
355 Two different types of measured FRFs (accelerance and receptance) from a shake table test are used in  
356 the updating process. The criterion for choosing appropriate frequency ranges has been discussed  
357 through this study. The results show that the FRFs of the updated model very closely match with the  
358 experimentally measured FRFs of the actual structure. Furthermore, a time-domain comparison between  
359 the simulated response and experimental response was conducted to verify the effectiveness of the  
360 model updating.

361 In the future, the FRF-based FE model updating will be performed on an actual space frame bridge,  
362 using field measurement data. More optimization algorithms will be studied for achieving better  
363 updating result.

364

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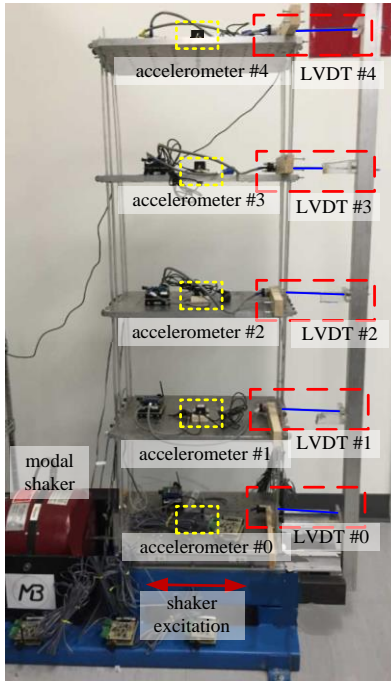


Figure 1. Frame structure with experimental setup

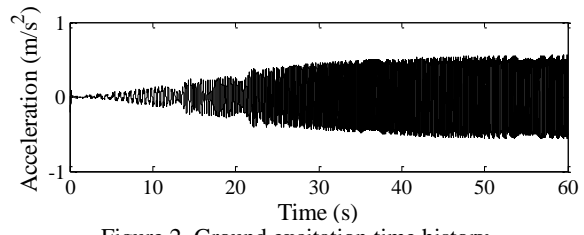


Figure 2. Ground excitation time history

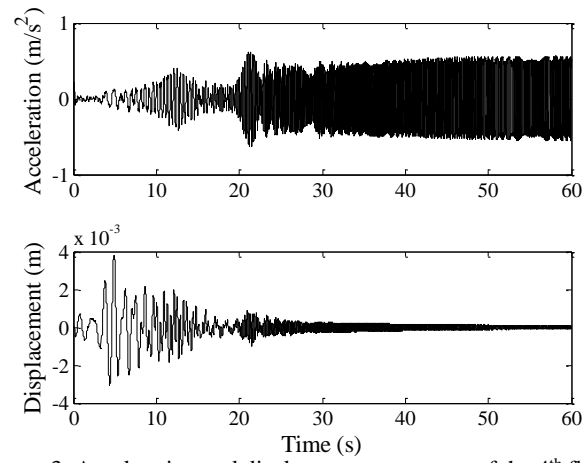


Figure 3. Acceleration and displacement responses of the 4<sup>th</sup> floor

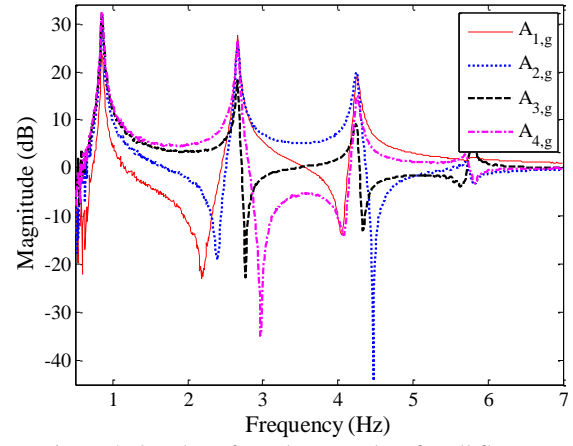


Figure 4. Overlay of acceleration plots for all floors

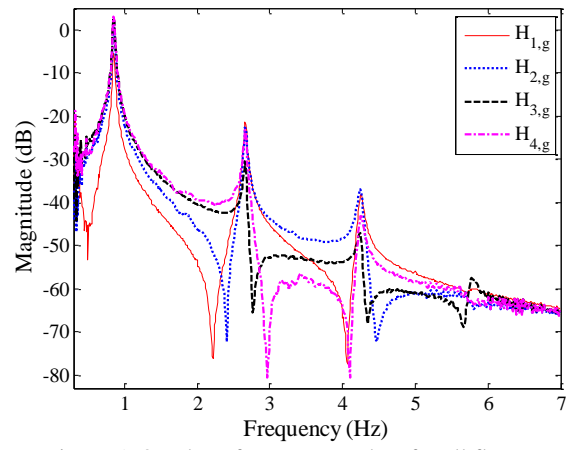
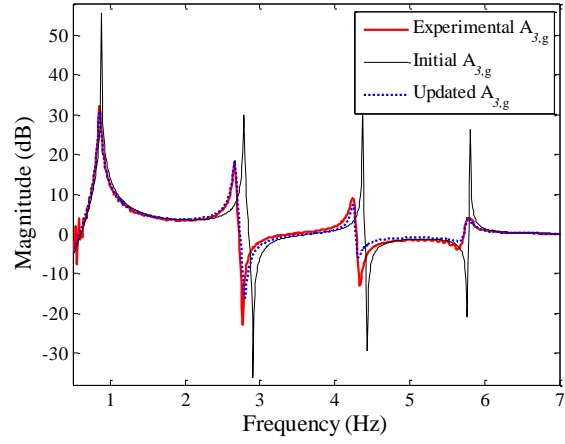
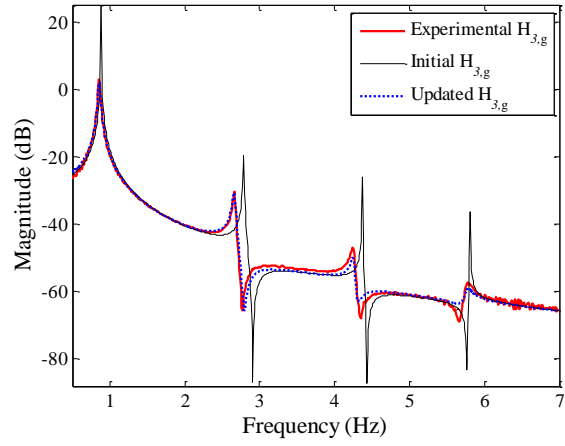


Figure 5. Overlay of receptance plots for all floors



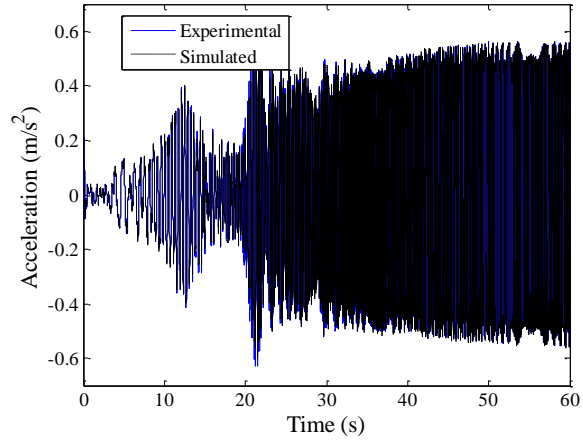
(a) Accelerance  $A_{3,g}$



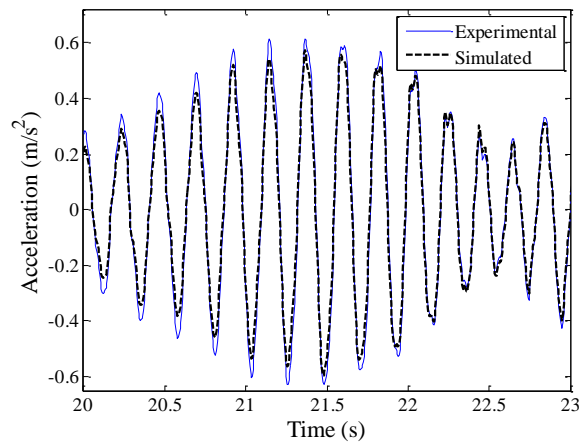
(b) Receptance  $H_{3,g}$

Figure 6. Comparison of the initial, the experimental and the updated FRFs



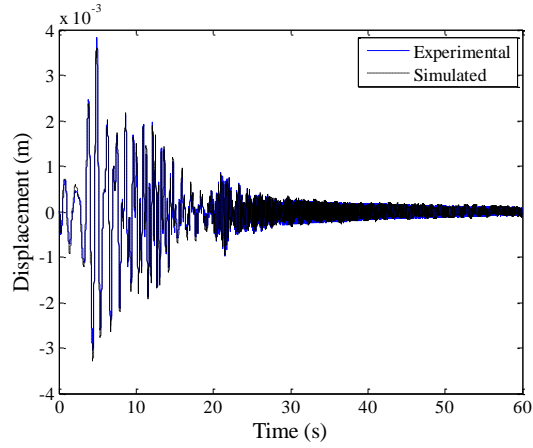


(a) Acceleration comparison on 4<sup>th</sup> floor

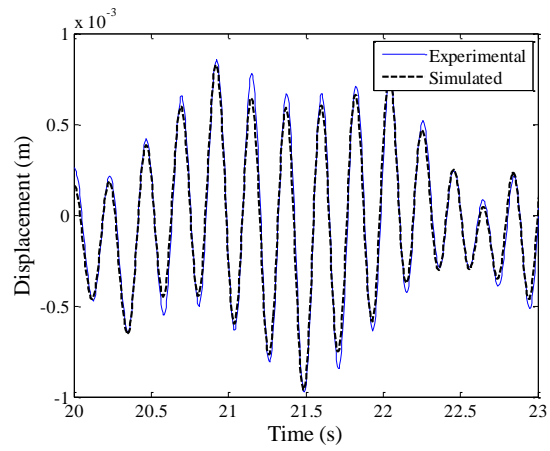


(b) Close-up acceleration comparison on 4<sup>th</sup> floor

Figure 7. Comparison between the experimental and simulated acceleration



(a) Displacement comparison on 4<sup>th</sup> floor



(b) Close-up displacement comparison on 4<sup>th</sup> floor

Figure 8 Comparison between the experimental and simulated displacement

**Table 1** Model updating results from shake table test.

Parameter (kg or N/m)	$m_1$	$m_2$	$m_3$	$m_4$	$k_1$	$k_2$	$k_3$	$k_4$
Initial	4.64	4.64	4.64	4.64	1019	1217	1420	2473
Updated (Accelerance)	5.16	5.14	4.94	5.14	1049.71	1282.53	1385.56	2732.56
Updated (Receptance)	5.16	5.16	4.94	5.14	1060.25	1285.84	1368.70	2743.77
Average of the updated	5.16	5.15	4.94	5.14	1054.98	1284.19	1377.13	2738.17